

Strengths and Weaknesses of Common Numerical Methods for Simulating Atmospheric Flows

Dale Durran
University of Washington

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Broad Categories

- Spatially discrete
 - Finite difference
 - Finite volume

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- Fluid dynamical viewpoint
 - Eulerian
 - Lagrangian
 - Semi-Lagrangian

Illustrative Equation

Transport in 2D nondivergent flow.

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- Flux form facilitates the construction of schemes with local (cell-wise) and global conservation.
- Advective form helps preserve uniform ψ in non-trivial velocity fields

Finite difference notation

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Local conservation:

$$\frac{d\phi_{i,j}}{dt} + \delta_x(u_{i,j}\phi_{i,j}) + \delta_y(v_{i,j}\phi_{i,j}) = 0$$

Finite-Difference Methods

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$$\left(\frac{d\psi}{dx}\right)_j = \delta_{2x}\phi_j + O[(\Delta x)^2], \quad \left(\frac{d\psi}{dx}\right)_j = \frac{4}{3}\delta_{2x}\phi_j - \frac{1}{3}\delta_{4x}\phi_j + O[(\Delta x)^4]$$

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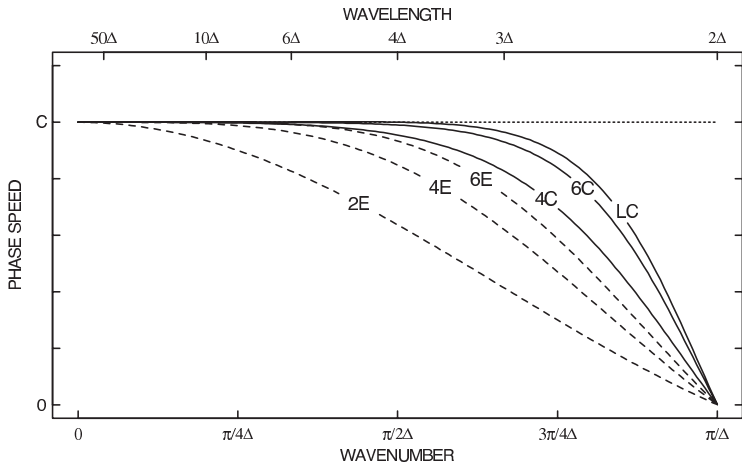
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- More advanced approach: a 4th-order compact scheme

$$\frac{1}{24} \left[5 \left(\frac{d\psi}{dx}\right)_{j+1} + 14 \left(\frac{d\psi}{dx}\right)_j + 5 \left(\frac{d\psi}{dx}\right)_{j-1} \right] = \frac{1}{12} (11\delta_{2x}\phi_j + \delta_{4x}\phi_j)$$

Phase Speed Error in 1D Advection

$$\text{Semi-discrete approximation to } \frac{\partial \psi}{\partial t} + c \frac{\partial \psi}{\partial x} = 0.$$



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- Convergence is never achieved in high Reynolds number atmospheric flow. (Why not?)
 - *Exception:* all those complicated linear solutions for nontrivial basic states!
- In atmospheric applications, errors are generally dominated by the most poorly resolved scales.
- High-order schemes *may* treat the marginally resolved scales better.

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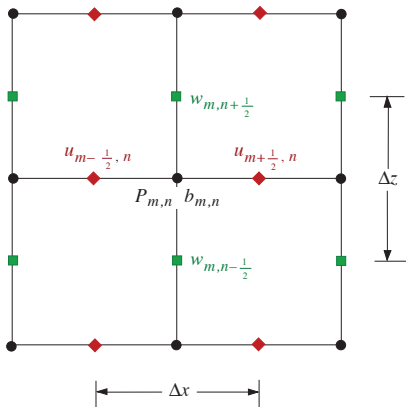
Utility of High Order II

Is it best to approximate all terms with differences having the same order of accuracy?

- Yes – if you are trying to achieve convergence.
- Not particularly if you are trying to improve the representation of poorly resolved scales.

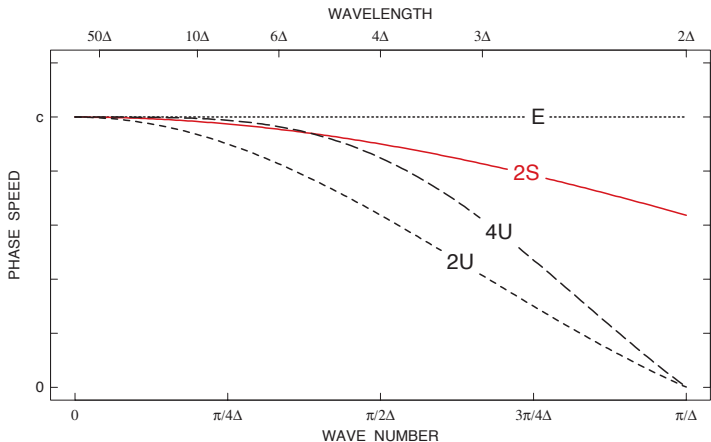
Example: Staggered Meshes

Arakawa C-grid



Example: Staggered Meshes II

Phase speeds using staggered (S) or unstaggered (U) 2nd- or 4th-order differences

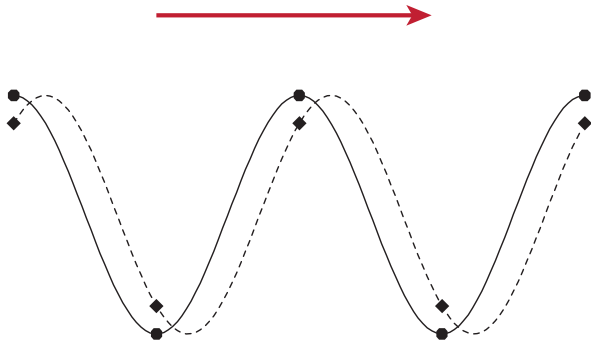


Trouble with $2\Delta x$ -Waves

Finite-difference methods do not propagate $2\Delta x$ -waves on an unstaggered mesh. Why not?

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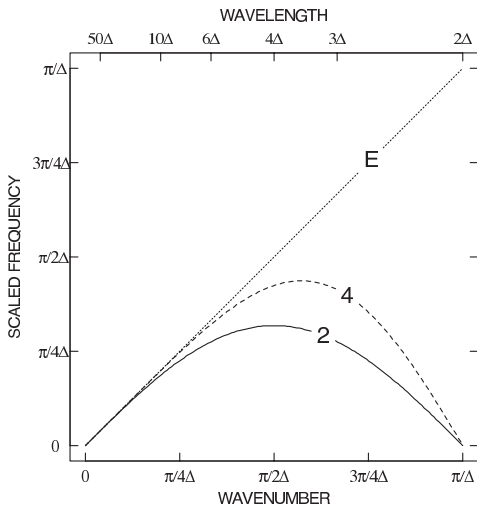


$2\Delta x$ -Waves and Negative Group Velocities

Animation of $2\Delta x$ -wide spike simulated by upstream and by explicit centered 2nd and 4th-order finite differences.

$2\Delta x$ -Waves and Negative Group Velocities

Group velocity is $\frac{\partial(\text{frequency})}{\partial(\text{wavenumber})}$



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 - Major focus of finite-volume methods

Finite Volume Methods

- Unknowns are *cell averages*:

$$\phi_j^n \approx \frac{1}{\Delta x} \int_{x_j - \Delta x/2}^{x_j + \Delta x/2} \psi(x, n\Delta t) dx$$

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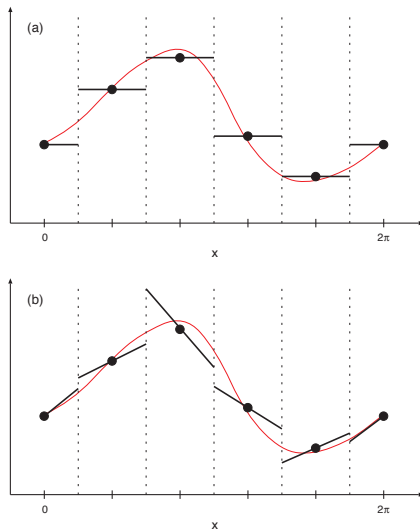
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- *Leads directly to two-level forward-in-time schemes*

Sub-Cell Reconstructions



Avoiding overshoots and undershoots

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$$\phi_j^{n+1} = H(\phi_{j-p}^n, \dots, \phi_{j+q+1}^n),$$

is *monotone* if

$$\frac{\partial H(\phi_{j-p}, \dots, \phi_{j+q+1})}{\partial \phi_i} \geq 0$$

for each integer i in the interval $[j - p, j + q + 1]$.

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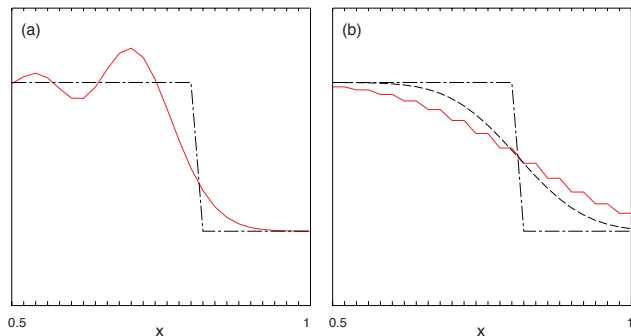
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Monotone schemes

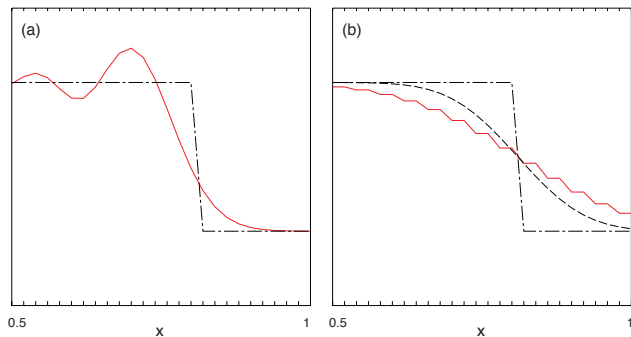
- Do not produce spurious ripples.
- Are first-order accurate.

Advection of a Step Function



- Left: 2nd-order centered with global 4th-derivative smoother
- Right: Upstream and Lax-Friedrichs (red) monotone methods

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Strategy: Scheme becomes monotone near discontinuities and is high-order elsewhere.

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When is it important in problems without shocks? *Chemically reacting flow*

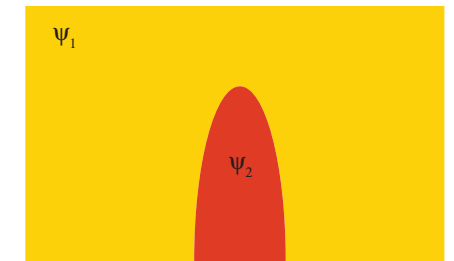
$$\begin{aligned}\frac{\partial \psi_1}{\partial t} + c \frac{\partial \psi_1}{\partial x} &= -r \psi_1 \psi_2, \\ \frac{\partial \psi_2}{\partial t} + c \frac{\partial \psi_2}{\partial x} &= r \psi_1 \psi_2.\end{aligned}$$

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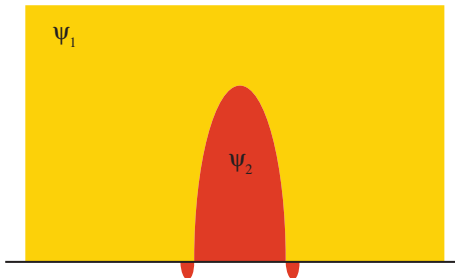


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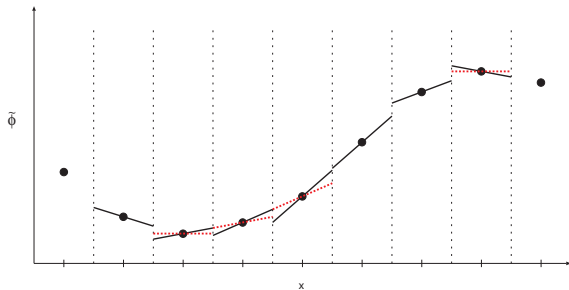
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Minmod limiter

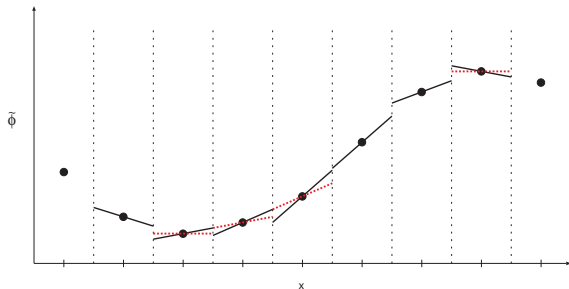
One local-smoothing strategy to prevent growth of ripples.



- If $\text{sgn}(\phi_{j+1} - \phi_j) \neq \text{sgn}(\phi_j - \phi_{j-1})$ slope in cell j is zero
- Otherwise $|\text{slope}|$ is $\min(|\phi_{j+1} - \phi_j|, |\phi_j - \phi_{j-1}|)$

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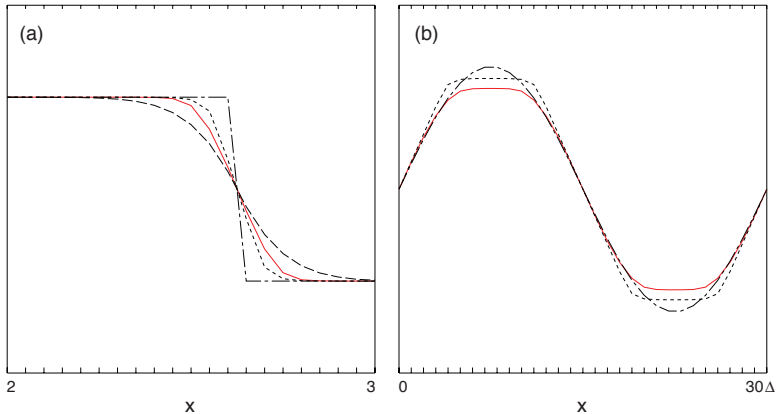


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Scheme is TVD, but not monotone.

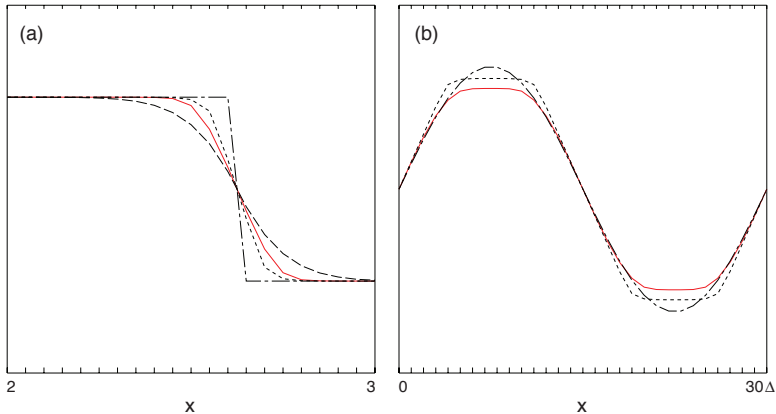
Limiters in Action

Minmod (long-dashed), MC (red), Superbee (short-dashed)



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Nice job at jump, but messes up smooth extremum.

Order of Accuracy

<i>Scheme</i>	<i>Estimated Order of Accuracy</i>
Upstream	0.9
Minmod Limiter	1.6
Superbee Limiter	1.6
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Table: Empirically determined order of accuracy for constant-wind-speed advection of a sine wave

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Avoid limiting smooth extrema!

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 - Global conservation of ϕ and ϕ^2
 - Collocation requirement that sets the residual to zero at a set of grid points
 - "Pseudo-spectral"
 - 50% faster than spectral
 - Lose conservation

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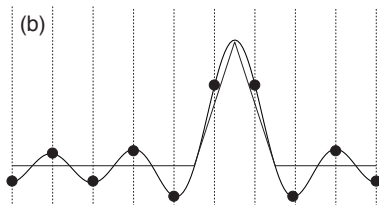
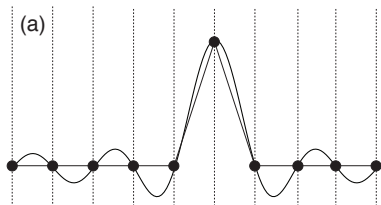
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- “Spectral accuracy” when approximating smooth functions — error goes to zero faster than any finite power of the effective grid spacing.
- Not conducive to preserving positivity or treating steep gradients

Global Overshoots and Undershoots with Poor Resolution



Global Spectral Model

- Expansion functions are spherical harmonics.
- Avoids problems with short time steps at poles.

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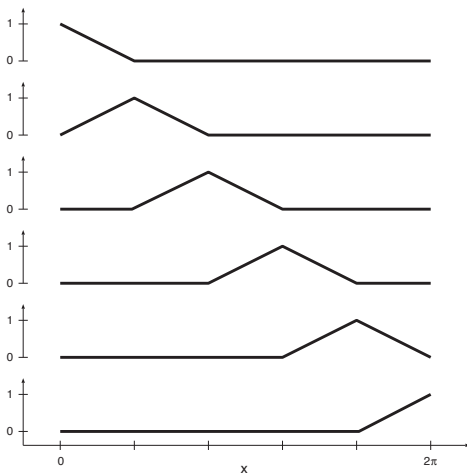
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- Common choice for the dynamical variables in global hydrostatic models.
- Moisture variables often finite volume or finite difference.

Finite Element Methods

Piecewise linear elements (chapeau functions)



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- Higher-order finite element methods are nevertheless the standard approach for solving many elliptic problems.

Finite Element versus Compact Differencing in 1D

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$$\frac{d}{dt} \left(\frac{a_{j+1} + 4a_j + a_{j-1}}{6} \right) + c \left(\frac{a_{j+1} - a_{j-1}}{2\Delta x} \right) = 0$$

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4th-order compact scheme:

$$\frac{d\phi_j}{dt} + c \left(\frac{d\psi}{dx} \right)_j = 0$$

$$\frac{1}{6} \left[\left(\frac{d\psi}{dx} \right)_{j+1} + 4 \left(\frac{d\psi}{dx} \right)_j + \left(\frac{d\psi}{dx} \right)_{j-1} \right] = \frac{\phi_{j+1} - \phi_{j-1}}{2\Delta x}$$

Finite Element versus Compact Differencing in 1D

Schemes are identical!

Finite Element versus Compact Differencing in 2D

4th-order compact scheme:

$$\frac{d\phi_{i,j}}{dt} + u \left(\frac{d\psi}{dx} \right)_{i,j} + v \left(\frac{d\psi}{dy} \right)_{i,j} = 0$$

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$$\frac{1}{6} \left[\left(\frac{d\psi}{dy} \right)_{i,j+1} + 4 \left(\frac{d\psi}{dy} \right)_{i,j} + \left(\frac{d\psi}{dy} \right)_{i,j-1} \right] = \frac{\phi_{i,j+1} - \phi_{i,j-1}}{2\Delta y}$$

Finite-elements: huge implicit mess (element couples with itself and 8 neighbors).

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How do we

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- Extend finite-element methods to high order while avoiding implicit coupling (and other bad behaviors)?
- Avoid the global coupling and $O(N^2)$ operation counts to update a set of expansion coefficients in spectral or pseudo-spectral methods?

One Solution

Use h - p methods – break domain into h elements and represent the solution within each element by orthogonal polynomials of maximum order p .

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- Spectral element methods – solution is continuous at cell boundaries (good for approximating 2nd-order derivatives).
- Discontinuous Galerkin methods – solution is discontinuous across cell boundaries (localizes communication between cells).

Discontinuous Galerkin Method

Enforce Galerkin criterion locally, within each element.

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Polynomial structure within each element is either

- *Modal* variant: Legendre polynomials
 - Orthogonal on $[-1, 1]$ with weight function unity

Discontinuous Galerkin Method

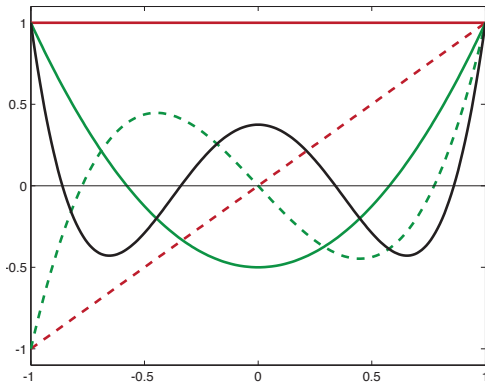
Enforce Galerkin criterion locally, within each element.

Polynomial structure within each element is either

- *Modal* variant: Legendre polynomials
 - Orthogonal on $[-1, 1]$ with weight function unity
- *Nodal* variant: Lagrange polynomials interpolating the Gauss-Legendre-Lobatto (GLL) points
 - Most accurate node placement for quadrature when there is a node at each end of the interval
 - Polynomials are not truly orthogonal
 - Discrete integrals are orthogonal due to the approximate quadrature on the GLL nodes

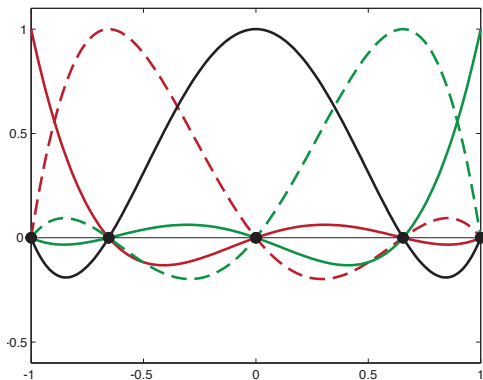
Modal DG

First 5 Legendre Polynomials



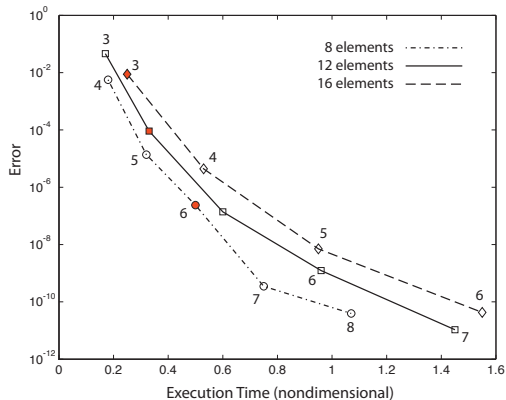
Nodal DG

Lagrange Polynomials Interpolating 5 GLL Nodes



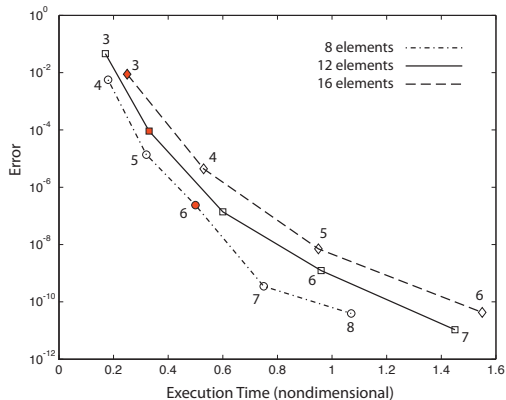
Convergence as a Function of Resolution

Nodal DG errors as a function of execution time



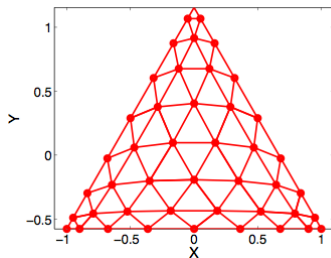
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Nodal DG errors as a function of execution time



Most efficient way to reduce error is to increase the polynomial order.

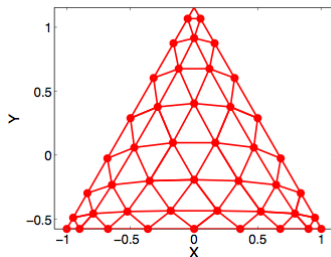
DG Considerations



Advantages:

- Suitable for massively parallel computing

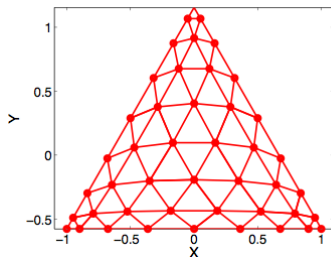
DG Considerations



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- Suitable for massively parallel computing
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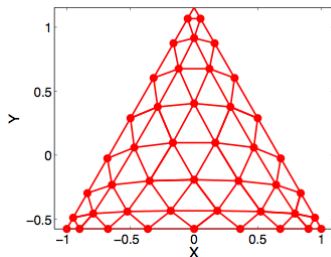
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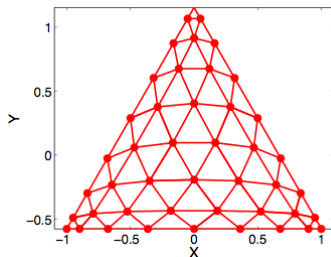
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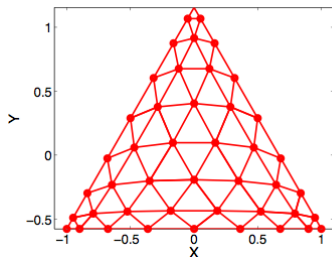
DG Considerations



Advantages:

- Suitable for massively parallel computing
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 - For high order polynomials, solution relatively insensitive to flux formulation
 - Lots of work to do within each element
- Rapid convergence for smooth solutions

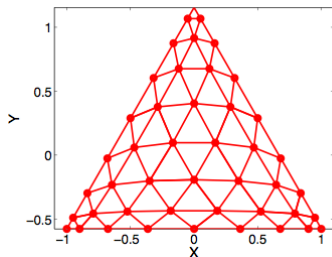
DG Considerations II



Disadvantages:

- Requires very short time step
 - Grid spacing reduced toward element boundaries

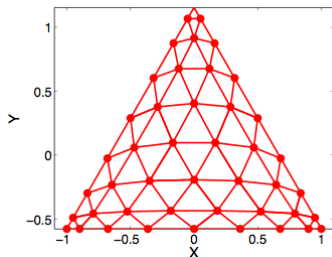
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- Discontinuities
 - Can be accommodated across element boundaries by limiting the fluxes
 - Cannot naturally be accommodated within each element

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Accelerations evaluated at fixed points.

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Accelerations evaluated along fluid parcel trajectories

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Fluid parcels arrive at every node on the specified mesh at the *end* of each time step.

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Semi-Lagrangian approximation to the advection equation is

$$\frac{\phi(x_j, t^{n+1}) - \phi(\tilde{x}_j^n, t^n)}{\Delta t} = 0,$$

where \tilde{x}_j^n denotes the departure point of a trajectory originating at time t^n and arriving at (x_j, t^{n+1}) .

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For constant $U > 0$

$$\tilde{x}_j^n = x_j - U\Delta t.$$

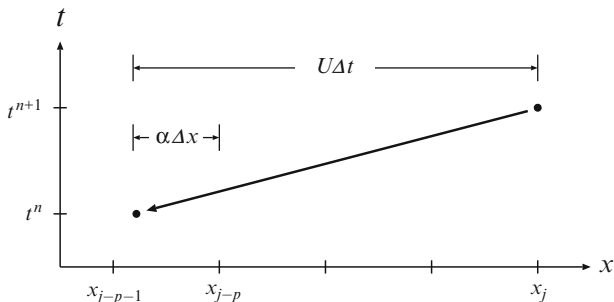
Semi-Lagrangian II

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Let p be the integer part of $U\Delta t/\Delta x$, then

$$x_{j-p-1} \leq \tilde{x}_j^n < x_{j-p}$$



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Interpolate the grid-point values to \tilde{x}_j^n .

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CFL Condition

Time step can be very large when simulating advection of a passive scalar (e.g., $U\Delta t/\Delta x = 4$). Do SL methods avoid the CFL condition?

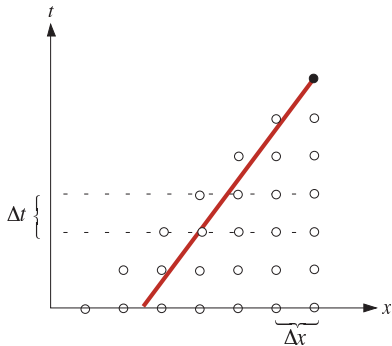
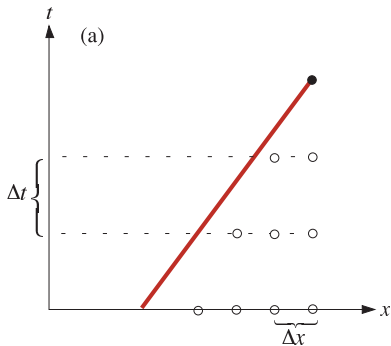
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Courant-Freidrichs-Levy Condition: the numerical domain of dependence must include the domain of dependence of the true solution. (Necessary but not sufficient for stability.)

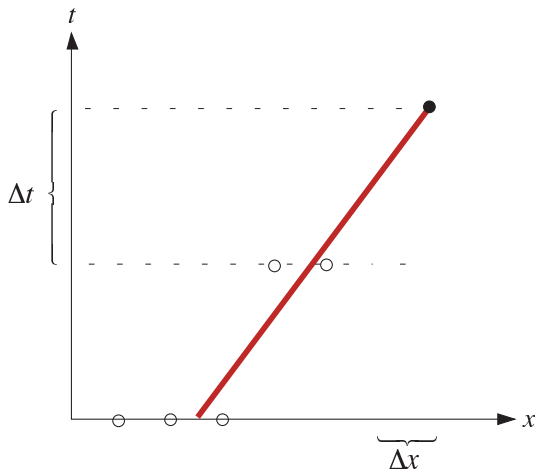
Eulerian CFL

Upstream differencing, constant-windspeed advection



Semi-Lagrangian CFL

Linear interpolation to departure point, constant-windspeed advection



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 - Trajectory morphology *not* simpler than the advected fields.

Reference

Durrant, D.R., 2010: *Numerical Methods for Fluid Dynamics: With Applications to Geophysics. 2nd Ed.* Springer.

